

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as *COM007* follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2020–19–07 Leonardo S.p.a.: Amendment 39–21250; Docket No. FAA–2020–0258; Product Identifier 2018–SW–002–AD.

(a) Applicability

This AD applies to Leonardo S.p.a. Model AW169 helicopters, certificated in any category, with left-hand (LH) main landing gear (MLG) assembly, part number (P/N) 6F3210V00132 or P/N 6F3210V00133, with serial number (S/N) MN01 through MN84 inclusive, or right-hand (RH) MLG assembly, P/N 6F3210V00232 or P/N 6F3210V00233, with S/N MN01 to MN81 inclusive, installed. This AD does not apply to helicopters with an MLG that has been modified in accordance with Magnaghi Aeronautica S.p.A. Service Bulletin No. SB–07–2017–AW169, Issue 5, dated November 22, 2017. This AD does not apply to MLG that have a “B” on the end of the serial number.

(b) Unsafe Condition

This AD defines the unsafe condition as an improperly tightened weight on wheels (WoW) support resulting in a rotation of the support and improper WoW switch performance, which if not corrected could lead to degraded attitude stabilization, and subsequent loss of control of the helicopter.

(c) Effective Date

This AD becomes effective October 27, 2020.

(d) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

(e) Required Actions

(1) Within 60 hours time-in-service, modify the WoW support installation as follows:

(i) Perform a short circuit connection between pin 26 of connector J343 and pin N of connector J319.

(ii) Cut the lockwire that locks the WoW microswitch and disconnect the WoW microswitch from the WoW support as depicted in Figure 1 of Leonardo Helicopters

Alert Service Bulletin No. 169–047, Revision A, dated February 19, 2018 (ASB 169–047).

(iii) Unscrew the nut and remove the washer and bolt. Remove from service the nut and bolt, but replace the washer.

(iv) Remove the WoW support from the MLG and remove any paint and clean areas where indicated by Figure 2 in ASB 169–047.

(v) Apply Alodine 1132 on cleaned areas of WoW support and landing gear strut leaving a 10 mm wide strip on the WoW support for structural glue EA934 application.

(vi) Apply a 10 mm wide strip of structural glue EA934 on the WoW support and install the WoW support on the MLG using a nut, bolt, and washer.

(vii) Torque the nut to 2.5 thru 3.5 Nm. Clean any excess glue and cure the glue on the structures for one hour at 60 °C/140 °F or eight days at room temperature (22 °C–26 °C/71.6 °F–78.8 °F).

(viii) Apply liquid jointing compound AMS–S–8802 Type 2 Class B, or equivalent, to the bolt and nut, as depicted in Figure 3 of ASB 169–047 and perform a microswitch adjustment for correct operation of the microswitch.

(ix) Mark the MLG nameplate by adding the letter “B” at the end of the S/N.

(x) Remove the short circuit connection between pin 26 of connector J343 and pin N of connector J319 as performed in paragraph (e)(1)(i) of this AD.

(2) After the effective date of this AD, do not install on any helicopter a LH or RH MLG assembly with a P/N and S/N listed in paragraph (a) of this AD unless it has been modified in accordance with the requirements of paragraph (e)(1) of this AD.

(f) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Rotorcraft Standards Branch, FAA, may approve AMOCs for this AD. Send your proposal to: Matt Fuller, AD Program Manager, Continued Operational Safety Branch, Airworthiness Products Section, General Aviation and Rotorcraft Unit, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone 817–222–5110; email 9-ASW-FTW-AMOC-Requests@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, the FAA suggests that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office, before operating any aircraft complying with this AD through an AMOC.

(g) Additional Information

The subject of this AD is addressed in European Aviation Safety Agency (now European Union Aviation Safety Agency) (EASA) No. 2017–0255, dated December 22, 2017. You may view the EASA AD on the internet at <https://www.regulations.gov> in Docket No. FAA–2020–0258.

(h) Subject

Joint Aircraft Service Component (JASC) Code: 3200, Landing Gear System.

(i) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Leonardo Helicopters Alert Service Bulletin No. 169–047, Revision A, dated February 19, 2018.

(ii) [Reserved]

(3) For Leonardo S.p.a. Helicopters service information identified in this AD, contact Leonardo S.p.a. Helicopters, Emanuele Bufano, Head of Airworthiness, Viale G. Agusta 520, 21017 C. Costa di Sarnate (Va) Italy; telephone +39–0331–225074; fax +39–0331–229046; or at <https://www.leonardocompany.com/en/home>.

(4) You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817–222–5110.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fedreg.legal@nara.gov, or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on September 9, 2020.

Gaetano A. Sciortino,

Deputy Director for Strategic Initiatives, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2020–20748 Filed 9–21–20; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2020–0342; Product Identifier 2019–SW–078–AD; Amendment 39–21242; AD 2020–19–01]

RIN 2120–AA64

Airworthiness Directives; Airbus Helicopters Deutschland GmbH Helicopters

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for all Airbus Helicopters Deutschland GmbH Model MBB–BK 117 D–2 helicopters. This AD was prompted by a report of an erroneous low rotor revolutions per minute (RPM) indication after establishing a one engine inoperative

(OEI) condition. This AD requires a software (SW) modification for the aircraft management computer (AMC). The FAA is issuing this AD to address the unsafe condition on these products. **DATES:** This AD is effective October 27, 2020.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of October 27, 2020.

ADDRESSES: For service information identified in this final rule, contact Airbus Helicopters, 2701 N Forum Drive, Grand Prairie, TX 75052; telephone 972-641-0000 or 800-232-0323; fax 972-641-3775; or at <https://www.airbus.com/helicopters/services/technical-support.html>. You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817-222-5110. It is also available on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0342.

Examining the AD Docket

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0342; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: George Schwab, Aviation Safety Engineer, Safety Management Section, Rotorcraft Standards Branch, FAA, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone 817-222-5110; email george.schwab@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to all Airbus Helicopters Deutschland GmbH Model MBB-BK 117 D-2 helicopters. The NPRM published in the **Federal Register** on April 28, 2020 (85 FR 23489). The NPRM was prompted by a report of an erroneous low rotor RPM indication after establishing an OEI condition. The NPRM proposed to require a SW modification for the AMC. The FAA is issuing this AD to address erroneous low RPM indications, which could cause the pilot to make inappropriate control inputs, resulting in damage to the helicopter or injury to occupants.

The European Union Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2019-0208, dated August 22, 2019 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Airbus Helicopters Deutschland GmbH Model MBB-BK 117 D-2 helicopters. You may examine the MCAI in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0342.

Comments

The FAA gave the public the opportunity to participate in developing

this final rule. The FAA received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

The FAA reviewed the relevant data and determined that air safety and the public interest require adopting this final rule as proposed, except for minor editorial changes. The FAA has determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM for addressing the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

Related Service Information Under 1 CFR Part 51

Airbus Helicopters has issued Alert Service Bulletin MBB-BK117 D-2-42A-005, Revision 3, dated June 6, 2019. This service information describes procedures for a SW modification for the AMC.

This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Differences Between This AD and the MCAI or Service Information

The MCAI provides a 60-day compliance time for accomplishing the SW modification. This AD requires completion of the SW modification within 50 hours time-in-service.

Costs of Compliance

The FAA estimates that this AD affects 30 helicopters of U.S. registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
1 work-hour × \$85 per hour = \$85	\$3,000	\$3,085	\$92,550

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section

44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
 (2) Will not affect intrastate aviation in Alaska, and
 (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2020–19–01 Airbus Helicopters

Deutschland GmbH: Amendment 39–21242; Docket No. FAA–2020–0342; Product Identifier 2019–SW–078–AD.

(a) Effective Date

This AD is effective October 27, 2020.

(b) Affected ADs

None.

(c) Applicability

This AD applies to all Airbus Helicopters Deutschland GmbH Model MBB–BK 117 D–2 helicopters, certificated in any category.

(d) Subject

Air Transport Association (ATA) of America Code 42, Integrated Modular Avionics.

(e) Reason

This AD was prompted by a report of an erroneous low rotor revolutions per minute (RPM) indication after establishing a one engine inoperative condition. The FAA is issuing this AD to address erroneous low RPM indications, which could cause the pilot to make inappropriate control inputs, resulting in damage to the helicopter or injury to occupants.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Definitions

(1) Affected part: An aircraft management computer (AMC) having a software (SW) version installed that is identified as “pre-modification SW” in Figure 1 to paragraphs (g)(1), (h), and (i) of this AD, or earlier SW version.

Figure 1 to Paragraphs (g)(1), (h), and (i) – Helicopter Configuration and Updated SW

Helicopter Configuration	Pre-modification SW	Post-modification/ Upgraded SW
D-2 and D-2m (basic)	As of the effective date of this AD, no D-2 and D-2m (basic) helicopters are known to be in service.	
D-2 and D-2m (Helionix Step 2)	V5.0.1 P/N D462C01S0501	V5.0.4 P/N D462C01S0504
	V5.0.2 P/N D462C01S0502	V5.0.4 P/N D462C01S0504
	V5.0.2 P/N D462C03S0502	V5.0.4 P/N D462C03S0504
D-2 and D-2m (Helionix Step 2.0.1)	V5.0.3 P/N D462C01S0503	V5.0.4 P/N D462C01S0504
	V5.0.3 P/N D462C03S0503	V5.0.4 P/N D462C03S0504
D-2 and D-2m (Helionix Step 3)	V6.0 P/N D462C01S0600	V6.0.2 P/N D462C01S0602
	V6.0 P/N D462C03S0600	V6.0.2 P/N D462C03S0602

(2) Group 1: Helicopters that have an affected part installed.

(3) Group 2: Helicopters that do not have an affected part installed.

(h) Software Modification

(1) For Group 1: Within 50 hours time-in-service after the effective date of this AD, update the SW of each affected part to the corresponding upgraded SW, as listed in Figure 1 to paragraphs (g)(1), (h), and (i) of this AD, in accordance with the Accomplishment Instructions, Section 3.B.2, of Airbus Helicopters Alert Service Bulletin MBB–BK117 D–2–42A–005, Revision 3, dated June 6, 2019.

(2) Replacement on a helicopter of an affected part with an AMC having the corresponding upgraded SW installed, as listed in Figure 1 to paragraphs (g)(1), (h), and (i) of this AD, or later SW upgrade is an

acceptable alternative method of compliance for the requirements of paragraph (h)(1) of this AD for that helicopter.

(i) Parts Installation Prohibition

Do not install on any helicopter an affected part, and do not upload any SW identified as “pre-modification SW” in Figure 1 to paragraphs (g)(1), (h), and (i) of this AD, or earlier SW version, on any AMC, as required by paragraph (i)(1) or (2) of this AD, as applicable.

(1) For Group 1: After modification of that helicopter as specified in paragraph (h) of this AD.

(2) For Group 2: As of the effective date of this AD.

(j) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Rotorcraft Standards Branch, FAA, may approve AMOCs for this AD. Send your proposal to: George Schwab, Aviation Safety Engineer, Safety Management Section, Rotorcraft Standards Branch, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone 817–222–5151; email 9-ASW-FTW-AMOC-Requests@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, notify your principal inspector or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office, before operating any aircraft complying with this AD through an AMOC.

(k) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information European Union Aviation Safety Agency (EASA) 2019–0208, dated August 22, 2019. This EASA AD may be found in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–0342.

(2) For service information identified in this AD, contact Airbus Helicopters, 2701 N Forum Drive, Grand Prairie, TX 75052; telephone 972–641–0000 or 800–232–0323; fax 972–641–3775; or at <https://www.airbus.com/helicopters/services/technical-support.html>. You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177.

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) Airbus Helicopters Alert Service Bulletin MBB–BK117 D–2–42A–005, Revision 3, dated June 6, 2019.

(ii) [Reserved]

(3) For service information identified in this AD, contact Airbus Helicopters, 2701 N Forum Drive, Grand Prairie, TX 75052; telephone 972–641–0000 or 800–232–0323; fax 972–641–3775; or at <https://www.airbus.com/helicopters/services/technical-support.html>.

(4) You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817–222–5110.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fedreg.legal@nara.gov, or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on September 1, 2020.

Gaetano A. Sciortino,

Deputy Director for Strategic Initiatives, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2020–20763 Filed 9–21–20; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2020–0793; Project Identifier MCAI–2020–00976–R; Amendment 39–21243; AD 2020–19–02]

RIN 2120–AA64

Airworthiness Directives; Airbus Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: The FAA is superseding Airworthiness Directive (AD) 2000–22–19 for Eurocopter France (now Airbus Helicopters) Model SA330F, G, and J helicopters. AD 2000–22–19 required repetitively inspecting certain tail rotor (T/R) blades for skin debonding and a crack. Since the FAA issued AD 2000–22–19, the inspection procedures have been revised. Additionally, the FAA is adding an affected part-numbered T/R blade and the FAA-validation for Model SA330F and G helicopters has been cancelled. This new AD revises the applicability, requires repetitively inspecting affected T/R blades with the new inspection procedures, and depending on the inspection results, repairing or replacing the T/R blade. This new AD also prohibits installing an affected T/R blade unless it has passed the inspections. The actions of this AD are intended to address an unsafe condition on these products.

DATES: This AD becomes effective October 7, 2020.

The Director of the Federal Register approved the incorporation by reference of a certain document listed in this AD as of October 7, 2020.

The FAA must receive comments on this AD by November 6, 2020.

ADDRESSES: You may send comments by any of the following methods:

- *Federal eRulemaking Docket:* Go to <https://www.regulations.gov>. Follow the online instructions for sending your comments electronically.
- *Fax:* 202–493–2251.
- *Mail:* Send comments to the U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590–0001.

• *Hand Delivery:* Deliver to the “Mail” address between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Examining the AD Docket

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–0793; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the European Aviation Safety Agency (EASA) (now European Union Aviation Safety Agency) AD, any service information that is incorporated by reference, any comments received, and other information. The street address for Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

For service information identified in this final rule, contact Airbus Helicopters, 2701 N Forum Drive, Grand Prairie, TX 75052; telephone 972–641–0000 or 800–232–0323; fax 972–641–3775; or at <https://www.airbus.com/helicopters/services/technical-support.html>. You may view the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. It is also available on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–0793.

FOR FURTHER INFORMATION CONTACT: Matt Fuller, AD Program Manager, Continued Operational Safety Branch, Airworthiness Products Section, General Aviation and Rotorcraft Unit, FAA, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone 817–222–5110; email matthew.fuller@faa.gov.

SUPPLEMENTARY INFORMATION:**Comments Invited**

This AD is a final rule that involves requirements affecting flight safety, and the FAA did not provide you with notice and an opportunity to provide your comments prior to it becoming effective. However, the FAA invites you to participate in this rulemaking by submitting written comments, data, or views. The most helpful comments reference a specific portion of the AD, explain the reason for any recommended change, and include supporting data. To ensure the docket does not contain duplicate comments, commenters should send only one copy of written comments, or if comments are filed electronically, commenters should submit them only one time.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will file in the docket all