

platforms may be required to gain proximity to the area being checked."

Corrective Actions

(b) Except as provided by paragraph (c) of this AD: If, during the inspection required by paragraph (a) of this AD, any corrosion or scratch is detected that is within the limits specified in Saab Service Bulletin 340-57-036, dated October 20, 1999, prior to further flight, perform corrective actions (including rework, an eddy current inspection, and repair) in accordance with Steps 2.B. and 2.C. of the Accomplishment Instructions of the service bulletin.

(1) If, after rework, the depth of the damage is less than or equal to 0.15 mm (0.006 inches) AND the damage does not exceed 15 percent of the area, no further action is required by this AD.

(2) If, after rework, the depth of the damage exceeds 0.15 mm (0.006 inches) but is less than or equal to 1.1 mm (0.043 inches), AND the damage does not exceed 30 percent of the area: Within 4,000 flight cycles after accomplishment of the inspection required by paragraph (a), repair the MLG support assembly in accordance with the service bulletin. Following the repair, no further action is required by this AD.

(c) If, during any inspection required by this AD, a discrepancy is detected for which the service bulletin specifies to contact Saab for appropriate action [including any crack or any corrosion or scratch that exceeds 1.1 mm (0.043 in) after applicable rework has been performed as required by paragraph (b) of this AD]: Prior to further flight, repair in accordance with a method approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate; or the Luftfartsverket (LFV) (or its delegated agent). For a repair method to be approved by the Manager, International Branch, ANM-116, as required by this paragraph, the Manager's approval letter must specifically reference this AD.

Alternative Methods of Compliance

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(f) Except as specified by paragraph (c) of this AD, the inspections and corrective

actions shall be done in accordance with Saab Service Bulletin 340-57-036, dated October 20, 1999. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Saab Aircraft AB, SAAB Aircraft Product Support, S-581.88, Linköping, Sweden. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 4: The subject of this AD is addressed in Swedish airworthiness directive 1-146, dated October 20, 1999.

(g) This amendment becomes effective on July 20, 2000.

Issued in Renton, Washington, on June 7, 2000.

Donald L. Riggan,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00-14951 Filed 6-14-00; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-56-AD; Amendment 39-11793; AD 2000-12-15]

RIN 2120-AA64

Airworthiness Directives; Dassault Model Falcon 2000, Mystere-Falcon 900, Falcon 900EX, Fan Jet Falcon, Mystere-Falcon 50, Mystere-Falcon 20, Mystere-Falcon 200, and Falcon 10 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to all Dassault Model Falcon 2000, Mystere-Falcon 900, Falcon 900EX, Fan Jet Falcon, Mystere-Falcon 50, Mystere-Falcon 20, Mystere-Falcon 200, and Falcon 10 series airplanes, that requires repetitive tests and inspections to detect discrepancies of the overwing emergency exit; and corrective action, if necessary. This amendment is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by this AD are intended to prevent failure of the overwing emergency exits to open, and consequent injury to passengers or crew members during an emergency evacuation.

EFFECTIVE DATE: July 20, 2000.

ADDRESSES: Information pertaining to this amendment may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT:

Norman B. Martenson, Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Dassault Model Falcon 2000, Mystere-Falcon 900, Falcon 900EX, Fan Jet Falcon, Mystere-Falcon 50, Mystere-Falcon 20, Mystere-Falcon 200, and Falcon 10 series airplanes was published in the **Federal Register** on April 24, 2000 (65 FR 21679). That action proposed to require repetitive tests and inspections to detect discrepancies of the overwing emergency exit; and corrective action, if necessary.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 767 airplanes of U.S. registry will be affected by this AD, that it will take approximately 1 work hour per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$46,020, or \$60 per airplane, per inspection cycle.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various

levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2000-12-15 Dassault Aviation:

Amendment 39-11793. Docket 2000-NM-56-AD.

Applicability: All Model Falcon 2000, Mystere-Falcon 900, Falcon 900EX, Fan Jet Falcon, Mystere-Falcon 50, Mystere-Falcon 20, Mystere-Falcon 200, and Falcon 10 series airplanes; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of the overwing emergency exits to open, and consequent injury to passengers or crew members during an emergency evacuation, accomplish the following:

Operational Test and Inspection

(a) Within 30 days after the effective date of this AD, perform an operational test and detailed visual inspection of the overwing emergency exit from inside the cabin to detect discrepancies (including separation, tearing, wearing, arcing, cracking) in the areas and components listed in Chapter 5 (ATA Code 52) of the applicable airplane maintenance manual (AMM). Accomplish the actions in accordance with the applicable AMM. If any discrepancy is detected during any test or inspection required by this paragraph, prior to further flight, repair in accordance with Chapter 5 (ATA Code 52) of the applicable AMM. Repeat the operational test and inspection thereafter at intervals not to exceed 13 months.

Note 2: For the purposes of this AD, a detailed visual inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required."

Alternative Methods of Compliance

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Note 4: The subject of this AD is addressed in French airworthiness directives 1999-213-025(B), 1999-212-024(B), 1999-211-023(B), and 1999-210-009(B); all dated May 19, 1999.

(d) This amendment becomes effective on July 20, 2000.

Issued in Renton, Washington, on June 9, 2000.

Donald L. Rigin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00-15188 Filed 6-14-00; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF THE TREASURY

Internal Revenue Service

26 CFR Part 1

[TD 8883]

RIN 1545-AW53

Guidance Under Section 1032 Relating to the Treatment of a Disposition by an Acquiring Entity of the Stock of a Corporation in a Taxable Transaction; Correction

AGENCY: Internal Revenue Service (IRS), Treasury.

ACTION: Correction to final regulations.

SUMMARY: This document contains a correction to Treasury Decision 8883, which was published in the **Federal Register** on Tuesday, May 16, 2000 (65 FR 31073), which provides guidance under section 1032 of the Internal Revenue Code, relating to the treatment of a disposition by an acquiring entity of the stock of a corporation in a taxable transaction.

EFFECTIVE DATE: This correction is effective May 16, 2000.

FOR FURTHER INFORMATION CONTACT: Filiz Serbes at (202) 622-7550 (not a toll-free number).

SUPPLEMENTARY INFORMATION:

Background

The final regulations that are the subject to correction are under section 1032 of the Internal Revenue Code.

Need for Correction

As published, Treasury Decision 8883 contains an error which may prove to be misleading and is in need of clarification.

Correction of Publication

Accordingly, the publication of the final regulations (TD 8883), which was the subject of FR Doc. 00-11900, is corrected as follows:

§ 1.1032-3 [Corrected]

On page 31076, column 3, § 1.1032-3(c)(1), is corrected to read as follows: