

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Boeing: Docket 2002–NM–198–AD.

Applicability: Model 767–200, –300, and –300F series airplanes, line numbers 1 through 883 inclusive; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To detect and correct cracks in the outer cylinder of the main landing gear (MLG), which could result in collapsed MLG and consequent reduced controllability of the airplane during takeoff and landing, accomplish the following:

Service Bulletin References

(a) The term “service bulletin,” as used in this AD, means the Accomplishment Instructions of Boeing Service Bulletin 767–32A0196, Revision 2, dated May 15, 2003.

Inspection and Corrective Actions

(b) Within 18 months after the effective date of this AD, for both MLG, do a gap measurement of the upper and lower joint gaps (includes measuring and recording upper and lower joint gaps twice); and an ultrasonic inspection of the outer cylinder of the main landing gear for cracks between the downlock fitting attach lugs, per Part 1 of the service bulletin.

(c) If no crack is found during the inspection required by paragraph (b) of this AD, before further flight, do the restoration (includes installing shims as applicable, electrical bracket, and cotter pins; and marking the main landing gear) per the service bulletin.

(d) If any crack is found during the inspection required by paragraph (b) of this AD: Before further flight, overhaul the outer cylinder of the MLG or replace the outer cylinder of the MLG with an interchangeable outer cylinder per Part 2 of the service bulletin, except as provided by paragraph (e) of this AD.

(e) If any crack is found in the outer cylinder that cannot be removed within the repair limits specified in the service bulletin, during the overhaul specified in paragraph (d) of this AD, and the service bulletin specifies to contact Boeing for appropriate action: Before further flight, repair per a method approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA; or per data meeting the type certification basis

of the airplane approved by a Boeing Company Designated Engineering Representative who has been authorized by the Manager, Seattle ACO, to make such findings. For a repair method to be approved, the approval must specifically reference this AD.

Note 1: When the outer cylinder is re-installed, attach the downlock fittings onto the outer cylinder as specified in the applicable Boeing Component Maintenance Manual (CMM), document number 161T1000, Section 32–11–19, temporary revision (TR) 32–61, dated March 26, 2002 or Section 32–11–19 pages 712 through 716 dated July 01, 2002, or dated July 01, 2003; or CMM 32–11–20, TR 32–62, dated March 26, 2002, or Section 32–11–20 pages 718 through 722 dated July 01, 2002, or dated July 01, 2003.

Actions Accomplished Per Previous Issue of Service Bulletin

(f) Accomplishment of the applicable actions before the effective date of this AD per Boeing Service Bulletin 767–32A0196, dated August 1, 2002; or, Revision 1, dated September 26, 2002; are considered acceptable for compliance with the corresponding action specified in this AD.

Parts Installation

(g) As of the effective date of this AD, no person may install a MLG on any airplane, unless the outer cylinder of the MLG has been inspected and follow-on and corrective actions have been accomplished per Boeing Service Bulletin 767–32A0196, Revision 2, dated May 15, 2003.

Alternative Methods of Compliance

(h) In accordance with 14 CFR 39.19, the Manager, Seattle Aircraft Certification Office, FAA, is authorized to approve alternative methods of compliance for this AD.

Issued in Renton, Washington, on November 28, 2003.

Kevin Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03–30220 Filed 12–4–03; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002–NM–174–AD]

RIN 2120–AA64

Airworthiness Directives; Boeing Model 737–300, –400, and –500 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the superseding of an existing airworthiness

directive (AD), applicable to certain Boeing Model 737 series airplanes. That AD currently requires a one-time general visual inspection of the seat locks and seat tracks of the flightcrew seats to ensure that the seats lock in position and to verify that lock nuts and bolts of adequate length are installed on the rear track lock bracket, and corrective action, if necessary. This action would revise the applicability of the existing AD by adding airplanes. The actions specified by the proposed AD are intended to prevent uncommanded movement of the flightcrew seats during acceleration and take-off of the airplane, which could result in reduced controllability of the airplane. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by January 20, 2004.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–114, Attention: Rules Docket No. 2002–NM–174–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227–1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain “Docket No. 2002–NM–174–AD” in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 or 2000 or ASCII text.

The service information referenced in the proposed rule may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124–2207. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. **FOR FURTHER INFORMATION CONTACT:** Shannon Lennon, Aerospace Engineer, Cabin Safety and Environmental Systems Branch, ANM–150S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington; telephone (425) 917–6435; fax (425) 917–6590.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall

identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2002-NM-174-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-174-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

On May 18, 2000, the FAA issued AD 2000-10-21, amendment 39-11745 (65 FR 34063, May 26, 2000), applicable to certain Boeing Model 737 airplanes, to require a one-time general visual inspection of the seat locks and seat tracks of the flightcrew seats to ensure that the seats lock in position and to verify that lock nuts and bolts of adequate length are installed on the rear track lock bracket, and corrective action, if necessary. That action was prompted by reports indicating instances of the pilot seat sliding to the aft-most position during acceleration and take-off on certain Boeing Model 737 series airplanes. The requirements of that AD

are intended to prevent uncommanded movement of the flightcrew seats during acceleration and take-off of the airplane, which could result in reduced controllability of the airplane.

Actions Since Issuance of Previous Rule

Since the issuance of that AD, the airplane manufacturer has informed the FAA that additional airplanes may be equipped with IPECO flightcrew seats that are subject to the identified unsafe condition. Those seats may have been installed on airplanes out of operators' stock or from an airplane not yet modified per AD 2000-10-21, amendment 39-11745.

Explanation of Relevant Service Information

We have reviewed and approved Boeing Alert Service Bulletin 737-25A1363, Revision 1, dated March 28, 2002, which revises the effectivity of the initial release (which was referenced as the appropriate source of service information for the actions required by AD 2000-10-21) of the service bulletin by adding additional airplanes. The procedures for a one-time general visual inspection and corrective actions if necessary specified in Revision 1 are essentially identical to those described in the initial release of the service bulletin. No more work is necessary on airplanes changed per the initial release of this service bulletin. Accomplishment of the actions specified in the alert service bulletin is intended to adequately address the identified unsafe condition.

Boeing Alert Service Bulletin 737-25A1363, Revision 1, dated March 28, 2002, refers to IPECO Service Bulletin A001-25-47, dated January 13, 1992, as an additional source of service information for accomplishment of the inspection of the seat locks and seat tracks of the flightcrew seats and corrective action, if necessary.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would supersede AD 2000-10-21 to continue to require a one-time general visual inspection of the seat locks and seat tracks of the flightcrew seats to ensure that the seats lock in position and to verify that lock nuts and bolts of adequate length are installed on the rear track lock bracket, and corrective action, if necessary. The proposed AD would also revise the applicability of the existing AD by adding airplanes. The actions would be required to be

accomplished in accordance with the Boeing alert service bulletin described previously, except as discussed below.

Difference Between Service Bulletin and Proposed AD

Operators should note that, although the Boeing alert service bulletin does not recommend a compliance time for accomplishment of the described actions, the FAA has determined that a 90-day compliance time would address the identified unsafe condition in a timely manner. In developing an appropriate compliance time for this proposed AD, we considered the degree of urgency associated with addressing the subject unsafe condition, the average utilization of the affected fleet, and the time necessary to perform the actions. In light of all of these factors, we find a 90-day compliance time for completion of the actions to be warranted, in that it represents an appropriate interval of time allowable for affected airplanes to continue to operate without compromising safety.

Explanation of Change Made to Existing Requirements

For clarification, we have revised the definition of a "general visual inspection" in this proposed AD.

Cost Impact

There are approximately 1,385 airplanes of the affected design in the worldwide fleet. The FAA estimates that 282 airplanes of U.S. registry would be affected by this proposed AD.

For Group 1 airplanes listed in Boeing Alert Service Bulletin 737-25A1363, Revision 1: The actions that are currently required by AD 2000-10-21 take approximately 3 work hours per airplane to accomplish, at an average labor rate of \$65 per work hour. Based on these figures, the cost impact of the currently required actions on U.S. operators is estimated to be \$195 per airplane.

For Group 2 airplanes listed in Boeing Alert Service Bulletin 737-25A1363, Revision 1: The new actions that are proposed in this AD action would take approximately 3 work hours per airplane to accomplish, at an average labor rate of \$65 per work hour. Based on these figures, the cost impact of the proposed requirements of this AD on U.S. operators is estimated to be \$195 per airplane.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the current or proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The

cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing amendment 39–11745 (65 FR

34063, May 26, 2000), and by adding a new airworthiness directive (AD), to read as follows:

Boeing: Docket 2002–NM–174–AD.
Supersedes AD 2000–10–21,
Amendment 39–11745.

Applicability: Model 737–300, –400, and –500 series airplanes equipped with IPECO flightcrew seats, as listed in Boeing Alert Service Bulletin 737–25A1363, Revision 1, dated March 28, 2002; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent uncommanded movement of the flightcrew seats during acceleration and take-off of the airplane, which could result in reduced controllability of the airplane, accomplish the following:

One-Time Inspection

(a) Perform a one-time general visual inspection of the seat locks and seat tracks of the flightcrew seats to ensure that the seats lock in position and to verify that lock nuts and bolts of adequate length are installed on the rear track lock bracket, at the applicable time and per the Work Instructions of the applicable service bulletin specified in Table 1 of this AD. Table 1 follows:

TABLE 1.—COMPLIANCE TIME/SERVICE BULLETIN

Airplanes—	Compliance Time—	Service Bulletin—
For Group 1 airplanes listed in Boeing Alert Service Bulletin 737–25A1363, Revision 1, dated March 28, 2002.	Within 90 days after September 26, 2001 (the effective date AD 2000–10–21, amendment 39–11745).	Boeing Alert Service Bulletin 737–25A1363, dated November 5, 1998.
For Group 2 airplanes listed in Boeing Alert Service Bulletin 737–25A1363, Revision 1, dated March 28, 2002.	Within 90 days after the effective date of this AD.	Boeing Alert Services Bulletin 737–25A1363, Revision 1, dated March 28, 2002.

Note 1: For the purposes of this AD, a general visual inspection is defined as: "A visual examination of an interior or exterior area, installation, or assembly to detect obvious damage, failure, or irregularity. This level of inspection is made from within touching distance unless otherwise specified. A mirror may be necessary to enhance visual access to all exposed surfaces in the inspection area. This level of inspection is made under normally available lighting conditions such as daylight, hangar lighting, flashlight, or droplight and may require removal or opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked."

Corrective Action

(1) If the seat lock pin fully engages in all lock positions of the seat track, and the rear track lock bracket is correctly installed: No further action is required by this AD.

(2) If the seat lock pin does not fully engage in all positions of the seat track, and lock nuts and bolts of adequate length are not installed on the rear track lock bracket: Prior to further flight, install lock nuts and bolts of adequate length on the track lock bracket

and verify proper seat movement and seat lock operation, in accordance with the applicable service bulletin.

Note 2: Boeing Alert Service Bulletin 737–25A1363, Revision 1, dated March 28, 2002, refers to IPECO Service Bulletin A001–25–47, dated January 13, 1992, as an additional source of service information for accomplishment of the actions required by paragraph (a) of this AD.

Actions Accomplished Per Previous Issue of Service Bulletin

(b) For Group 2 airplanes: Inspections and corrective actions accomplished before the effective date of this AD per Boeing Alert Service Bulletin 737–25A1363, dated November 5, 1998, are considered acceptable for compliance with the corresponding actions specified in this AD.

Alternative Methods of Compliance

(c)(1) In accordance with 14 CFR 39.19, the Manager, Seattle Aircraft Certification Office (ACO), FAA, is authorized to approve alternative methods of compliance (AMOCs) for this AD.

(2) Alternative methods of compliance, approved previously per AD 2000–10–21, amendment 39–11745, are approved as alternative methods of compliance with the requirements of this AD.

Issued in Renton, Washington, on November 28, 2003.

Kevin Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
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