received, the direct final rule would become final on the date noted above. The NRC did not receive any comments that warranted withdrawal of the direct final rule. Therefore, this rule will become effective as scheduled.

Dated at Rockville, Maryland, this 20th day of December 2001.

For the Nuclear Regulatory Commission. **Michael T. Lesar.**

Chief, Rules and Directives Branch, Division of Administrative Services, Office of Administration.

[FR Doc. 01–31923 Filed 12–27–01; 8:45 am] BILLING CODE 7590–01–P

FEDERAL RESERVE SYSTEM

12 CFR Parts 208 and 225

[Regulations H and Y; Docket No. R-1055]

Risk–Based Capital Guidelines; Capital Adequacy Guidelines; Capital Maintenance; Capital Treatment of Recourse, Direct Credit Substitutes and Residual Interests in Asset Securitizations

AGENCY: Board of Governors of the Federal Reserve System **ACTION:** Final rule; correction

SUMMARY: This document corrects the Federal Reserve's regulatory text of a final rule published in the **Federal Register** of November 29, 2001 (66 FR 59614), regarding the capital treatment of recourse, direct credit substitutes, and residual interests in asset securitizations. This correction rectifies errors made in Attachment II in Appendix A, part 208 and Appendix A, part 225.

DATES: This correction is effective January 1, 2002.

FOR FURTHER INFORMATION CONTACT: Thomas Boemio, 202–452–2982 or Arleen Lustig, 202–452–2987.

SUPPLEMENTARY INFORMATION:

Correction:

In the final rule FR Doc. No. 01– 29179, beginning on 66 FR 59614 in the issue of November 29, 2001, make the following corrections.

PART 208—[CORRECTED]

1. In Appendix A to Part 208, Attachment II, on page 59643:

A. In the column for Components, in the fourth entry under Supplementary Capital, replace the word "stocks" with the word "stock."

B. In the column for Minimum requirements, the fourth entry is revised to read, "Banks should avoid using minority interests to introduce elements not otherwise qualifying for tier 1 capital."

C. In the column for Minimum requirements, remove the eleventh entry beginning with "As a general rule * * *" in its entirety.

D. Remove footnote 3 following the table.

PART 225—[CORRECTED]

2. In Appendix A to Part 225, Attachment II, on page 59651:

A. In the column for Minimum requirements, the second entry is revised to read "Organizations should avoid using minority interests to introduce elements not otherwise qualifying for tier 1 capital."

B. In the column for Minimum requirements, in the eleventh entry of the table, replace the word "banks" with "organizations."

By order of the Board of Governors of the Federal Reserve System, December 20, 2001.

Jennifer J. Johnson,

Secretary of the Board.

[FR Doc. 01–31887 Filed 12–27–01; 8:45 am] BILLING CODE 31887-22–S

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001–NE–40–AD; Amendment 39–12569; AD 2001–26–05]

RIN 2120-AA64

Airworthiness Directives; Hamilton Sundstrand Model 247F Propellers

AGENCY: Federal Aviation Administration, DOT. **ACTION:** Final rule; request for comments.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that is applicable to certain Hamilton Sundstrand model 247F propellers. This action requires a one-time rework of certain model 247F propellers by removing all four propeller blades from service, replacing those blades with serviceable propeller blades, and marking the propeller with a new part number. This amendment is prompted by nine reports of the blades partially slipping at the bond joint between the blade tulip and the composite blade airfoil interface. The actions specified in this AD are intended to prevent the loss of a propeller blade, which may result in loss of airplane control.

DATES: Effective January 14, 2002. The incorporation by reference of certain

publications listed in the rule is approved by the Director of the Federal Register as of January 14, 2002.

Comments for inclusion in the Rules Docket must be received on or before February 26, 2002.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 2001-NE-40-AD, 12 New England Executive Park, Burlington, MA 01803–5299. Comments may be inspected at this location, by appointment, between 8:00 a.m. and 4:30 p.m., Monday through Friday, except Federal holidays. Comments may also be sent via the Internet using the following address: 9-aneadcomment@faa.gov. Comments sent via the Internet must contain the docket number in the subject line. The service information referenced in this AD may be obtained from Hamilton Sundstrand. A United Technologies Company, Publications Manager, Mail Stop 2AM-EE50, One Hamilton Road, Windsor Locks, CT 06096. This information may be examined, by appointment, at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA, or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Frank Walsh, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803–5299; telephone (781) 238–7158, fax (781) 238–7199.

SUPPLEMENTARY INFORMATION: The FAA has received nine reports of blades partially slipping at the bond joint between the blade tulip and the composite blade airfoil interface on Hamilton Sundstrand model 247F propellers, part numbers (P/N's) 810610-1 and 815550-1. Investigation reveals that this partial slippage is due to debonding of that interface. This amendment requires, within 30 days of the effective date of this AD as a onetime action, reworking certain model 247F propellers by removing all four existing propeller blades P/N's R810640-1, R810640-2, and R810640-3 from service, replacing those blades with serviceable propeller blades, and marking the propeller with a new part number. To date, no blade has come free from the hub. This condition, if not corrected, could result in the loss of a propeller blade, which may result in loss of airplane control.