

and anticipates making every effort to distribute the same scenarios to all covered banks.

In addition to the minimum three scenarios, the FDIC may require a covered bank with significant trading activities to include factors related to trading and counterparty risk in its stress test. Typically, these factors might include additional shocks to specific market prices, interest rates, rate spreads, or other key market variables consistent with historical or hypothetical adverse market events.

IV. Development and Distribution

As one part of the process of developing scenarios, the FDIC will gather information from outside entities and develop themes for the stress test scenarios, including the identification of potentially material vulnerabilities or salient risks to the financial system, and consider potential paths for individual variables. The outside entities may include academic experts, staffs of international organizations, foreign supervisors, financial institutions that regularly provide forecasts, and other private sector risk analysts that regularly conduct stress tests based on U.S. and global economic and financial scenarios. The FDIC will use the information gathered in this manner to inform its consideration of potential risks and scenarios.

The Office of the Comptroller of the Currency ("OCC"), the Board of Governors of the Federal Reserve System ("Board"), and the FDIC (collectively, the "Agencies") expect to consult closely to develop scenarios for stress testing. Absent specific supervisory concerns, the FDIC anticipates that the annual stress test scenarios distributed by the FDIC will be the same as or nearly identical to the scenarios developed by the Board for the supervisory stress tests conducted by the Board under Section 165(i)(1). This would mean the same economic and financial variables following the same paths as used in the scenarios for the Board's supervisory stress tests.

Although the Agencies generally expect to consult closely on scenario development, they may have different views of risks that should be reflected in the stress test scenarios used by covered banks for the annual stress test. The FDIC may distribute scenarios to covered banks that differ in certain respects from those distributed by the OCC and the Board if necessary to better reflect specific FDIC concerns. The FDIC expects such situations to be extremely rare, however, and anticipates making every effort to avoid differences in the scenarios required by each agency.

The FDIC anticipates that the stress test scenarios will be revised annually as appropriate to ensure that each scenario remains relevant under prevailing economic and industry conditions. These yearly revisions will enable the scenarios to capture evolving risks and vulnerabilities. The need to ensure that scenarios do not become outdated because of economic and financial developments makes a lengthy process of review and comment concerning scenarios prior to distribution each year impractical. However, the process of consultation with the Board and the OCC, as well as the ongoing interaction of FDIC staff with public and private sector experts to obtain views on salient risks and to obtain suggestions for the behavior of key economic variables, should ensure that the stress conditions reflected in the scenarios are well suited to their purpose.

The scenario development process culminates with the distribution of the scenarios to all covered banks no later than November 15th of each year. The scenario descriptions provided to covered banks will include values for economic and financial variables depicting the paths those variables follow under the scenarios. The FDIC believes that distribution of the scenarios no later than November 15th aligns with similar processes at the OCC and the Board.

Federal Deposit Insurance Corporation.

Dated at Washington, DC, this 14th day of November 2012.

Valerie J. Best,

Assistant Executive Secretary.

[FR Doc. 2012-28104 Filed 11-19-12; 8:45 am]

BILLING CODE 6714-01-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2012-0528; Directorate Identifier 2011-SW-068-AD; Amendment 39-17261; AD 2012-23-05]

RIN 2120-AA64

Airworthiness Directives; Eurocopter Deutschland Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for all Eurocopter Deutschland GmbH (ECD) Model MBB-BK117 C-2 helicopters with certain Generator Control Units

(GCU) installed. This AD requires replacing the GCUs. This AD was prompted by reports of internal short circuits in certain GCUs. These actions are intended to prevent a short circuit, which could result in a loss of electrical generating power, loss of systems required for continued safe flight and landing, and subsequent loss of control of the helicopter.

DATES: This AD is effective December 26, 2012.

ADDRESSES: For service information identified in this AD, contact American Eurocopter Corporation, 2701 N. Forum Drive, Grand Prairie, TX 75052, telephone (972) 641-0000 or (800) 232-0323, fax (972) 641-3775, or at <http://www.eurocopter.com/techpub>. You may review the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the Docket Operations Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, any incorporated-by-reference service information, the economic evaluation, any comments received, and other information. The street address for the Docket Operations Office (phone: 800-647-5527) is U.S. Department of Transportation, Docket Operations Office, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

George Schwab, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Safety Management Group, 2601 Meacham Blvd., Fort Worth, Texas 76137; telephone (817) 222-5114; email george.schwab@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

On May 22, 2012, at 77 FR 30230, the **Federal Register** published our notice of proposed rulemaking (NPRM), which proposed to amend 14 CFR part 39 to include an AD that would apply to ECD Model MBB-BK117 C-2 helicopters with a GCU, part number (P/N) 51530-021EI with no modification (MOD), MOD A, or MOD B installed. That NPRM proposed to require replacing the GCU within 300 hours time-in-service (TIS) or 6 months, whichever occurred earlier. The proposed requirements were intended to prevent loss of electrical generating power, resulting in the loss

of systems required for continued safe flight and landing, and subsequent loss of control of the helicopter.

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA AD No. 2011-0149R1, dated September 30, 2011 (AD 2011-0149R1), to correct an unsafe condition for the Eurocopter Deutschland GmbH Model MBB-BK117 C-2 helicopters. EASA advises that during an acceptance test procedure of a GCU, a short circuit caused by a manufacturing discrepancy occurred within the unit. According to EASA, all P/N 51530-021EI "no MOD," "MOD A," and "MOD B" GCUs are potentially affected by this discrepancy. To address this potential unsafe condition, EASA issued AD No. 2011-0149, dated August 19, 2011, to identify and replace each affected GCU with an airworthy GCU. Since issuing that AD, ECD demonstrated that helicopters modified in accordance with ECD Alert Service Bulletin (ASB) MBB BK117 C-2-24A-008, Revision 1, dated August 29, 2011, have a much lower risk of losing electrical generating power from a faulty generator control unit. EASA then revised AD No. 2011-0149 and issued AD 2011-0149R1 to allow an extended compliance time for helicopters modified in accordance with the ECD ASB.

Comments

We gave the public the opportunity to participate in developing this AD, but we did not receive any comments on the NPRM (77 FR 30230, May 22, 2012).

FAA's Determination

These helicopters have been approved by the aviation authority of Germany and are approved for operation in the United States. Pursuant to our bilateral agreement with Germany, EASA, its technical representative, has notified us of the unsafe condition described in the EASA AD. We are issuing this AD because we evaluated all information provided by EASA and determined the unsafe condition exists and is likely to exist or develop on other helicopters of these same type designs and that air safety and the public interest require adopting the AD requirements as proposed.

Differences Between This AD and the EASA AD

The EASA AD allows a compliance time of 1,500 flight hours for helicopters previously modified by ECD ASB MBB BK117 C-2-24A-008 Revision 1, while this AD requires compliance within the next 300 hours TIS or 6 months,

whichever occurs first, for all affected helicopters.

Related Service Information

ECD has issued ASB MBB-BK117 C-2-24A-010 Revision 2, dated September 14, 2011, which specifies removing any GCU with P/N 51530-021EI with no modification (MOD), MOD A, or MOD B, and replacing it with a GCU P/N 51530-021EI MOD C or later MOD. EASA classified this ASB as mandatory and issued AD 2011-0149R1 to ensure the continued airworthiness of these helicopters.

Costs of Compliance

We estimate that this AD will affect 104 helicopters of U.S. Registry. We estimate that operators may incur the following costs in order to comply with this AD. Replacing a GCU with an airworthy GCU will require about 2 work hours at an average labor rate of \$85 per hour. Required parts will cost \$7,130, for a total cost per helicopter of \$7,300.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on helicopters identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

(1) Is not a "significant regulatory action" under Executive Order 12866;

(2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);

(3) Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction; and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared an economic evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2012-23-05 Eurocopter Deutschland

GMBH: Amendment 39-17261; Docket No. FAA-2012-0528; Directorate Identifier 2011-SW-068-AD.

(a) Applicability

This AD applies to Model MBB-BK117 C-2 helicopters with a generator control unit (GCU), part number (P/N) 51530-021EI with no modification (MOD), MOD A, or MOD B installed, certificated in any category.

(b) Unsafe Condition

This AD defines the unsafe condition as an internal short circuit in certain GCUs. This condition could result in loss of electrical generating power, resulting in the loss of systems required for continued safe flight and landing, and subsequent loss of control of the helicopter.

(c) Effective Date

This AD becomes effective December 26, 2012.

(d) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

(e) Required Actions

(1) Within the next 300 hours time-in-service or 6 months, whichever occurs first, replace all GCUs with no MOD, MOD A, or MOD B with an airworthy GCU.

(2) Do not install a GCU P/N 51530-021-EI with no MOD, MOD A, or MOD B on any helicopter.

(f) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Safety Management Group, FAA, may approve AMOCs for this AD. Send your proposal to: George Schwab, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Safety Management Group, 2601 Meacham Blvd., Fort Worth, Texas 76137; telephone (817) 222-5114; email george.schwab@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office, before operating any aircraft complying with this AD through an AMOC.

(g) Additional Information

(1) Eurocopter Deutschland GmbH Alert Service Bulletin MBB-BK117 C-2-24A-010 Revision 2, dated September 14, 2011, which is not incorporated by reference, contains additional information about the subject of this AD.

(2) For service information identified in this AD, contact American Eurocopter Corporation, 2701 N. Forum Drive, Grand Prairie, TX 75052, telephone (972) 641-0000 or (800) 232-0323, fax (972) 641-3775, or at <http://www.eurocopter.com/techpub>. You may review a copy of the service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137.

(3) The subject of this AD is addressed in European Aviation Safety Agency AD No. 2011-0149R1, dated September 30, 2011.

(h) Subject

Joint Aircraft Service Component (JASC)
Code: 2436: DC Generator Control Unit.

Issued in Fort Worth, Texas, on October 30, 2012.

Kim Smith,

Directorate Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 2012-28039 Filed 11-19-12; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2012-1188; Directorate Identifier 2012-SW-049-AD; Amendment 39-17254; AD 2012-10-53]

RIN 2120-AA64

Airworthiness Directives; Eurocopter Deutschland GmbH Helicopters

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule; request for comments.

SUMMARY: We are adopting a new airworthiness directive (AD) for Eurocopter Deutschland GmbH (ECD) Model EC135 P1, EC135 P2, EC135 P2+, EC135 T1, EC135 T2, and EC135 T2+ helicopters. This is the **Federal Register** publication of an Emergency AD (EAD) that was previously sent to all known owners and operators of these helicopters. That EAD superseded an earlier related EAD. This AD requires, before further flight and at specified intervals, checking and inspecting the upper and lower main rotor hub (MRH) shaft flanges for a crack, and inspecting the lower hub-shaft flange bolt attachment areas for a crack. This AD is prompted by three reported incidents of cracking on the lower hub-shaft flanges of EC135 model helicopters. These actions are intended to detect a crack on the hub-shaft flange, which if not corrected could result in failure of the MRH and subsequent loss of control of the helicopter.

DATES: This AD becomes effective December 5, 2012 to all persons except those persons to whom it was made immediately effective by Emergency AD No. 2012-10-53, issued on May 18, 2012, which contained the requirements of this AD.

We must receive comments on this AD by January 22, 2013.

ADDRESSES: You may send comments by any of the following methods:

- **Federal eRulemaking Docket:** Go to <http://www.regulations.gov>. Follow the online instructions for sending your comments electronically.
- **Fax:** 202-493-2251.
- **Mail:** Send comments to the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590-0001.
- **Hand Delivery:** Deliver to the "Mail" address between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the Docket Operations Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the economic evaluation, any comments received, and other information. The street address for the Docket Operations Office (telephone 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

For service information identified in this AD, contact American Eurocopter Corporation, 2701 N. Forum Drive, Grand Prairie, Texas 75052; telephone (972) 641-0000 or (800) 232-0323; fax (972) 641-3775; or at <http://www.eurocopter.com/techpub>. You may review the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137.

FOR FURTHER INFORMATION CONTACT: Gary Roach, Aviation Safety Engineer, Regulations and Policy Group, Rotorcraft Directorate, FAA, 2601 Meacham Blvd., Fort Worth, Texas 76137; telephone (817) 222-5110; email gary.b.roach@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

This AD is a final rule that involves requirements affecting flight safety, and we did not provide you with notice and an opportunity to provide your comments prior to it becoming effective. However, we invite you to participate in this rulemaking by submitting written comments, data, or views. We also invite comments relating to the economic, environmental, energy, or federalism impacts that resulted from adopting this AD. The most helpful comments reference a specific portion of the AD, explain the reason for any recommended change, and include supporting data. To ensure the docket does not contain duplicate comments, commenters should send only one copy of written comments, or if comments are filed electronically, commenters should submit them only one time. We will file in the docket all comments that we receive, as well as a report summarizing each substantive public contact with FAA personnel concerning this rulemaking during the comment period. We will consider all the comments we receive and may conduct additional rulemaking based on those comments.

Discussion

On May 15, 2012, we issued Emergency AD 2012-10-51 for the ECD Model EC135 series helicopters to detect a crack on the MRH shaft flange. Emergency AD 2012-10-51 required a pilot check of the lower MRH shaft flange for a crack or deformed blade attachment bolt safety pins before the first flight of each day, inspecting the upper and lower MRH shaft flanges for a crack within 5 hours time-in-service (TIS), and replacing the MRH shaft if there is a crack.

After we issued Emergency AD 2012-10-51, the European Aviation Safety