

regulatory burden. Accordingly, this proposed rule will not have a significant economic impact on a substantial number of small credit unions, and therefore, no regulatory flexibility analysis is required.

Paperwork Reduction Act

NCUA has determined that this rule will not increase paperwork requirements under the Paperwork Reduction Act of 1995 and regulations of the Office of Management and Budget.

Executive Order 13132

Executive Order 13132 encourages independent regulatory agencies to consider the impact of their actions on state and local interests. In adherence to fundamental federalism principles, NCUA, an independent regulatory agency as defined in 44 U.S.C. 3502(5), voluntarily complies with the executive order. This proposed rule would not have a substantial direct effect on the states, on the relationship between the national government and the states, or on the distribution of power and responsibilities among the various levels of government. NCUA has determined that this proposed rule does not constitute a policy that has federalism implications for purposes of the executive order.

The Treasury and General Government Appropriations Act, 1999—Assessment of Federal Regulations and Policies on Families

NCUA has determined that this proposed rule will not affect family well-being within the meaning of section 654 of the Treasury and General Government Appropriations Act, 1999, Public Law 105–277, 112 Stat. 2681 (1998).

Agency Regulatory Goal

NCUA's goal is to promulgate clear and understandable regulations that impose minimal regulatory burden. We request your comments on whether this proposed rule is understandable and minimally intrusive if implemented as proposed.

List of Subjects

12 CFR Part 701

Credit unions.

12 CFR Part 723

Credit, Credit unions, Reporting and recordkeeping requirements.

12 CFR Part 742

Credit unions, Reporting and recordkeeping requirements.

By the National Credit Union Administration Board on March 18, 2010.

Mary Rupp,

Secretary of the Board.

For the reasons discussed above, NCUA proposes to amend 12 CFR parts 701, 723, and 742 as follows:

PART 701—ORGANIZATION AND OPERATIONS OF FEDERAL CREDIT UNIONS

1. The authority citation for part 701 continues to read as follows:

Authority: 12 U.S.C. 1752(5), 1755, 1756, 1757, 1759, 1761a, 1761b, 1766, 1767, 1782, 1784, 1787, and 1789. Section 701.6 is also authorized by 31 U.S.C. 3717. Section 701.31 is also authorized by 15 U.S.C. 1601 *et seq.*, 42 U.S.C. 1861 and 42 U.S.C. 3601–3610. Section 701.35 is also authorized by 42 U.S.C. 4311–4312.

2. Amend § 701.36 by revising paragraphs (d) introductory text and (d)(1) to read as follows:

§ 701.36 FCU ownership of fixed assets.

* * * * *

(d) *Regulatory Flexibility Program.* Federal credit unions that meet Regulatory Flexibility Program standards, as determined pursuant to Part 742 of this chapter, are exempt from the three-year partial occupancy requirement described in paragraph (b) of this section when acquiring unimproved land for future expansion pursuant to the terms of section 742.4(a)(3) of this chapter. For a Federal credit union eligible for the Regulatory Flexibility Program that subsequently loses eligibility:

(1) Section 742.3 of this chapter provides that NCUA may require the credit union to divest any existing fixed assets for substantive safety and soundness reasons; and

* * * * *

PART 723—MEMBER BUSINESS LOANS

3. The authority citation for part 723 continues to read as follows:

Authority: 12 U.S.C. 1756, 1757, 1757A, 1766, 1785, 1789.

§ 723.7 [Amended]

4. Amend § 723.7 by removing the last sentence of paragraph (b).

PART 742—REGULATORY FLEXIBILITY PROGRAM

5. The authority citation for part 742 continues to read as follows:

Authority: 12 U.S.C. 1756, 1766.

§ 742.4 [Amended]

6. Amend § 742.4 by removing the first sentence of paragraph (a)(3) and by removing paragraphs (a)(4), (a)(5), and (a)(6) and redesignating paragraphs (a)(7), (a)(8), and (a)(9) as paragraph (a)(4), (a)(5), and (a)(6).

[FR Doc. 2010–6391 Filed 3–24–10; 8:45 am]

BILLING CODE 7535–01–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2010–0217; Directorate Identifier 2009–NE–23–AD]

RIN 2120–AA64

Airworthiness Directives; Pratt & Whitney (PW) PW4000 Series Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for PW PW4052, PW4056, PW4060, PW4062, PW4062A, PW4074, PW4077, PW4077D, PW4084D, PW4090, PW4090–3, PW4152, PW4156, PW4156A, PW4158, PW4164, PW4168, PW4168A, PW4460, and PW4462 turbofan engines. This proposed AD would require initial and repetitive fluorescent penetrant inspections (FPI) for cracks in the blade locking and loading slots of the high-pressure compressor (HPC) drum rotor disk assembly. This proposed AD results from reports of cracked locking and loading slots in the HPC drum rotor disk assembly. We are proposing this AD to detect cracks in the locking and loading slots in the HPC drum rotor disk assemblies, which could result in rupture of the HPC drum rotor disk assembly and damage to the airplane.

DATES: We must receive any comments on this proposed AD by May 24, 2010.

ADDRESSES: Use one of the following addresses to comment on this proposed AD.

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- *Mail:* Docket Management Facility, U.S. Department of Transportation, 1200 New Jersey Avenue, SE., West Building Ground Floor, Room W12–140, Washington, DC 20590–0001.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5

p.m., Monday through Friday, except Federal holidays.

- *Fax:* (202) 493-2251.

Contact Pratt & Whitney, 400 Main St., East Hartford, CT 06108; telephone (860) 565-7700; fax (860) 565-1605, for information identified in this proposed AD.

FOR FURTHER INFORMATION CONTACT: Rose Len, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: rose.len@faa.gov; telephone (781) 238-7772; fax (781) 238-7199.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send us any written relevant data, views, or arguments regarding this proposal. Send your comments to an address listed under **ADDRESSES**. Include "Docket No. FAA-2010-0217; Directorate Identifier 2009-NE-23-AD" in the subject line of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of the Web site, anyone can find and read the comments in any of our dockets, including, if provided, the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review the DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78).

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is the same as the Mail address provided in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

Discussion

We have received reports of 294 HPC drum rotor disk assemblies that were removed because of cracks in the rear drum locking and loading slots. We determined that the cracks resulted from thermal mechanical fatigue. Cracks in rotating life-limited parts (LLPs), such as the HPC rear drum of the HPC drum rotor disk assembly, could result in rupture of that part.

Relevant Service Information

We have reviewed and approved the technical contents of PW Service Bulletins (SBs) PW4ENG 72-796, dated June 11, 2009, PW4G-100-72-186, Revision 1, dated September 2, 2004, and PW4G-112-72-264, Revision 1, dated September 2, 2004, that describe procedures for performing a local FPI of the HPC drum rotor disk assembly blade locking and loading slots for cracks.

FAA's Determination and Requirements of the Proposed AD

We have evaluated all pertinent information and identified an unsafe condition that is likely to exist or develop on other products of this same type design. We are proposing this AD, which would require performing a repetitive local FPI for cracks in the HPC drum rotor disk assembly blade locking and loading slots. The proposed AD would require you to use the service information described previously to perform these actions.

Interim Action

These actions are interim actions and we may take further rulemaking actions in the future.

Costs of Compliance

We estimate that this proposed AD would affect 1,038 engines installed on airplanes of U.S. registry. We also estimate that it would take about 1 work-hour per engine to perform the proposed actions, and that the average labor rate is \$85 per work-hour. No parts are required. Based on these figures, we estimate the total cost of the proposed AD to U.S. operators to be \$88,230.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that

section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Would not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD. You may get a copy of this summary at the address listed under **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive:

Pratt & Whitney: Docket No. FAA-2010-0217; Directorate Identifier 2009-NE-23-AD.

Comments Due Date

- (a) The Federal Aviation Administration (FAA) must receive comments on this

airworthiness directive (AD) action by May 24, 2010.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Pratt & Whitney (PW) PW4052, PW4056, PW4060, PW4062, PW4062A, PW4074, PW4077, PW4077D, PW4084D, PW4090, PW4090-3, PW4152, PW4156, PW4156A, PW4158, PW4164, PW4168, PW4168A, PW4460, and PW4462 turbofan engines. These engines are installed on, but not limited to, Boeing 747-400, 767-

200, 767-300, 777-200, and 777-300 airplanes; McDonnell Douglas MD-11 airplanes; and Airbus A300-600, A310-300, and A330-200 airplanes.

Unsafe Condition

(d) This AD results from reports of cracked locking and loading slots in the high-pressure compressor (HPC) drum rotor disk assembly. We are issuing this AD to detect cracks in the locking and loading slots in the HPC drum rotor disk assemblies, which could result in rupture of the HPC drum rotor disk assembly and damage to the airplane.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified unless the actions have already been done.

Local Fluorescent Penetrant Inspection

(f) Perform a local fluorescent penetrant inspection for cracks in the HPC drum rotor disk assembly blade locking and loading slots of the specific stages of the HPC drum rotor disk assemblies from which any of the blades are removed as specified in Table 1 of this AD.

TABLE 1—COMPLIANCE TIMES AND SERVICE BULLETINS BY ENGINE MODEL

For engine model	Inspect whenever—	Use—
(1) PW4074, PW4077, PW4077D, PW4084D, PW4090, and PW4090-3.	Any of the 13th or 14th stage blades are removed during a shop visit.	Paragraphs 1.A. through 1.B. of the Accomplishment Instructions of PW4G-112-72-264, Revision 1, dated September 2, 2004.
(2) PW4164, PW4168, and PW4168A	Any of the 13th, 14th, or 15th stage blades are removed during a shop visit.	Paragraphs 1.A. through 1.C. of the Accomplishment Instructions of PW4G-100-72-186, Revision 1, dated September 2, 2004.
(3) PW4052, PW4056, PW4060, PW4062, PW4062A, PW4152, PW4156, PW4156A, PW4158, PW4460, and PW4462.	Any of the 13th, 14th, or 15th stage blades are removed during a shop visit.	Paragraphs 1.A. through 1.C. of the Accomplishment Instructions of PW4ENG 72-796, dated June 11, 2009.

(g) Remove from service any HPC drum rotor disk assembly found with a crack in the blade loading and locking slots of the HPC drum rotor disk assembly.

Alternative Methods of Compliance

(h) The Manager, Engine Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

Related Information

(i) Contact Rose Len, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: rose.len@faa.gov; telephone (781) 238-7772; fax (781) 238-7199, for more information about this AD.

(j) Pratt & Whitney Service Bulletins PW4ENG 72-796, dated June 11, 2009, PW4G-100-72-186, Revision 1, dated September 2, 2004, and PW4G-112-72-264, Revision 1, dated September 2, 2004, pertain to the subject of this AD. Contact Pratt & Whitney, 400 Main St., East Hartford, CT 06108; telephone (860) 565-7700; fax (860) 565-1605, for a copy of this service information.

Issued in Burlington, Massachusetts, on March 16, 2010.

Francis A. Favara,

Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 2010-6581 Filed 3-24-10; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2008-1095; Directorate Identifier 2008-NE-34-AD]

RIN 2120-AA64

Airworthiness Directives; Pratt & Whitney (PW) Model PW2037, PW2037(M), and PW2040 Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for PW Model PW2037, PW2037(M), and PW2040 turbofan engines. This proposed AD would require removing erosion damage on fan blades with cutback leading edges and restoring the leading edge contour. This proposed AD results from reports from PW that fan blade leading edge erosion can result in a fan thrust deterioration mode (FTDM) condition, which reduces the engine's capability of producing full rated take-off thrust. We are proposing this AD to prevent loss of engine thrust from an FTDM condition, which could result in an inability to maintain safe flight.

DATES: We must receive any comments on this proposed AD by May 24, 2010.

ADDRESSES: Use one of the following addresses to comment on this proposed AD.

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- *Mail:* Docket Management Facility, U.S. Department of Transportation, 1200 New Jersey Avenue, SE., West Building Ground Floor, Room W12-140, Washington, DC 20590-0001.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

- *Fax:* (202) 493-2251.

You can get the service information identified in this proposed AD from Pratt & Whitney, 400 Main St., East Hartford, CT 06108; telephone (860) 565-8770; fax (860) 565-4503.

FOR FURTHER INFORMATION CONTACT:

Mark Riley, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: mark.riley@faa.gov; telephone (781) 238-7758; fax (781) 238-7199.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send us any written relevant data, views, or arguments regarding this proposal. Send your comments to an address listed under **ADDRESSES**. Include "Docket No. FAA-2008-1095; Directorate Identifier 2008-NE-34-AD" in the subject line of your comments. We specifically invite