

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, Washington 98124-2207; telephone 206-544-5000, extension 1; fax 206-766-5680; e-mail me.boecom@boeing.com; Internet <https://www.myboeingfleet.com>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on December 30, 2010.

Suzanne Masterson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2011-371 Filed 1-11-11; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2010-0228; Directorate Identifier 2009-NM-252-AD; Amendment 39-16574; AD 2011-02-01]

RIN 2120-AA64

Airworthiness Directives; The Boeing Company Model MD-11 and MD-11F Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD requires a one-time inspection to detect damage of the wire assemblies of the tail tank fuel system, a wiring change, and corrective actions if necessary. This AD also requires, for certain airplanes, a general visual inspection for correct installation of the self-adhering high-temperature electrical insulation tape; installation of a wire assembly support bracket and routing wire assembly; changing wire supports; and installation

of a wire protection bracket. This AD was prompted by fuel system reviews conducted by the manufacturer. We are issuing this AD to detect and correct a potential of ignition sources inside fuel tanks, which, in combination with flammable vapors, could result in a fuel tank fire or explosion, and consequent loss of the airplane.

DATES: This AD is effective February 16, 2011.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of February 16, 2011.

ADDRESSES: For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, 3855 Lakewood Boulevard, MC D800-0019, Long Beach, California 90846-0001; telephone 206-544-5000, extension 2; fax 206-766-5683; e-mail dse.boecom@boeing.com; Internet <https://www.myboeingfleet.com>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (phone: 800-647-5527) is Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Serj Harutunian, Aerospace Engineer, Propulsion Branch, ANM-140L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712-4137; phone: (562) 627-5254; fax: (562) 627-5210; e-mail: Serj.Harutunian@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a supplemental notice of proposed rulemaking (SNPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to the specified products. That SNPRM published in the **Federal Register** on November 17, 2010 (75 FR 70150). The original NPRM (75 FR 12464, March 16, 2010) proposed to require a one-time inspection to detect damage of the wire assemblies of the tail tank fuel system, a wiring change, and corrective actions if necessary. The SNPRM proposed to revise the original NPRM by adding, for certain airplanes, a general visual inspection for correct installation of the self-adhering high-temperature electrical insulation tape; installation of a wire assembly support bracket and routing wire assembly; changing wire supports; and installation of a wire protection bracket.

Comments

We gave the public the opportunity to participate in developing this AD. We have considered the comment received. FedEx supports the SNPRM.

Explanation of Change Made to the AD

We have revised this AD to identify the legal name of the manufacturer as published in the most recent type certificate data sheet for the affected airplane models.

Conclusion

We reviewed the relevant data, considered the comment received, and determined that air safety and the public interest require adopting the AD as proposed except for minor editorial changes and the change described previously. We have determined that these minor changes:

- Are consistent with the intent that was proposed in the SNPRM for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the SNPRM.

We also determined that these changes will not increase the economic burden on any operator or increase the scope of the AD.

Costs of Compliance

We estimate that this AD affects 110 airplanes of U.S. registry.

We estimate the following costs to comply with this AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Inspection of tail tank fuel system wire assembly.	1 work-hour × \$85 per hour = \$85	\$0	\$85	\$9,350.
Inspection of electrical insulation tape	1 work-hour × \$85 per hour = \$85	0	85	\$9,350.
Change wire supports	3 work-hours × \$85 per hour = \$255	9	264	Up to \$29,040.

We estimate the following costs to do any necessary installations and repairs that would be required based on the

results of the inspection. We have no way of determining the number of

aircraft that might need these installations and repairs.

ON-CONDITION COSTS

Action	Labor cost	Parts cost	Cost per product
Installation/repair	Up to 23 work-hours × \$85 per hour = \$1,955	\$11,829	Up to \$13,784.
Adjust tape installation	1 × \$85 per hour = \$85	0	\$85.

According to the manufacturer, some of the costs of this AD may be covered under warranty, thereby reducing the cost impact on affected individuals. We do not control warranty coverage for affected individuals. As a result, we have included all costs in our cost estimate.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701: "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),
- (3) Will not affect intrastate aviation in Alaska, and

- (4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

- Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2011-02-01 The Boeing Company:
Amendment 39-16574 ; Docket No. FAA-2010-0228; Directorate Identifier 2009-NM-252-AD.

Effective Date

- (a) This AD is effective February 16, 2011.

Affected ADs

- (b) None.

Applicability

(c) This AD applies to The Boeing Company Model MD-11 and MD-11F airplanes, certificated in any category, as identified in Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010.

Subject

(d) Joint Aircraft System Component (JASC)/Air Transport Association (ATA) of America Code 28: Fuel.

Unsafe Condition

(e) This AD was prompted by fuel system reviews conducted by the manufacturer. We are issuing this AD to detect and correct a potential of ignition sources inside fuel tanks, which, in combination with flammable vapors, could result in a fuel tank fire or explosion, and consequent loss of the airplane.

Compliance

(f) Comply with this AD within the compliance times specified, unless already done.

Action

(g) For airplanes in Group 1, Configuration 1; and Group 2, Configuration 1: Within 60 months after the effective date of this AD, perform a general visual inspection to detect damage of wire assemblies of the tail tank fuel system, in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010.

(1) For airplanes in Group 1, Configuration 1: If no damage is found, before further flight, apply self-adhering high-temperature electrical insulation tape on the wire assemblies, install wire assembly support brackets, route wire assemblies, install extruded channel wire supports, and install a wire protection bracket, in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010.

(2) For airplanes in Group 1, Configuration 1: If damage is found, before further flight, repair or replace the wire assemblies, apply

self-adhering high-temperature electrical insulation tape on the wire assemblies, install wire assembly support brackets, route wire assemblies, install extruded channel wire supports, and install a wire protection bracket, in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010.

(3) For airplanes in Group 2, Configuration 1: If no damage is found, before further flight, install wire assembly support brackets, route wire assemblies, install extruded channel wire supports, and install a wire protection bracket, in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010.

(4) For airplanes in Group 2, Configuration 1: If damage is found, before further flight, repair or replace the wire assemblies, install wire assembly support brackets, route wire assemblies, install extruded channel wire supports, and install a wire protection bracket, in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010.

(h) For airplanes in Group 1, Configuration 2: Within 60 months after the effective date of this AD, do a general visual inspection for correct installation of the self-adhering high-temperature electrical insulation tape, and change the wire supports, in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010. If the self-adhering high-temperature electrical insulation tape is installed incorrectly, before further flight, adjust the tape installation to achieve the correct dimensions, in accordance with Figure 1 of Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010.

(i) For airplanes in Group 2, Configuration 2: Within 60 months after the effective date of this AD, change the wire supports, in accordance with Figure 2 of Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010.

Alternative Methods of Compliance (AMOCs)

(j)(1) The Manager, Los Angeles Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in the Related Information section of this AD.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

Related Information

(k) For more information about this AD, contact Serj Harutunian, Aerospace Engineer, Propulsion Branch, ANM-140L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California

90712-4137; phone: (562) 627-5254; fax: (562) 627-5210; e-mail: Serj.Harutunian@faa.gov.

Material Incorporated by Reference

(l) You must use Boeing Alert Service Bulletin MD11-28A124, Revision 1, dated August 24, 2010, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information, under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, 3855 Lakewood Boulevard, MC D800-0019, Long Beach, California 90846-0001; telephone 206-544-5000, extension 2; fax 206-766-5683; e-mail dse.boecom@boeing.com; Internet <https://www.myboeingfleet.com>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at an NARA facility, call 202-741-6030, or go to http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on January 3, 2011.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2011-271 Filed 1-11-11; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2010-0225; Directorate Identifier 2009-NM-203-AD; Amendment 39-16525; AD 2010-24-06]

RIN 2120-AA64

Airworthiness Directives; Short Brothers PLC Model SD3 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are superseding an existing airworthiness directive (AD) that applies to the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) issued by an airworthiness authority of another country to identify

and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as

Subsequent to accidents involving Fuel Tank System explosions in flight * * * and on ground, * * * Special Federal Aviation Regulation 88 (SFAR88) * * * required a safety review of the aircraft Fuel Tank System * * *.

* * * * *

Fuel Airworthiness Limitations are items arising from a systems safety analysis that have been shown to have failure mode(s) associated with an 'unsafe condition' * * *. These are identified in Failure Conditions for which an unacceptable probability of ignition risk could exist if specific tasks and/or practices are not performed in accordance with the manufacturers' requirements.

* * * * *

We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective February 16, 2011.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of February 16, 2011.

The Director of the Federal Register approved the incorporation by reference of certain other publications listed in this AD as of July 21, 2006 (71 FR 34801, June 16, 2006).

ADDRESSES: You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC.

FOR FURTHER INFORMATION CONTACT: Todd Thompson, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1175; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a supplemental notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That supplemental NPRM was published in the **Federal Register** on August 4, 2010 (75 FR 46864), and proposed to supersede AD 2006-12-18, Amendment 39-14644 (71 FR 34801, June 16, 2006). That NPRM proposed to require revising the airplane flight manual (AFM); revising the Airworthiness Limitation (AWL) section; doing a resistance check, inspection, and jumper installation; and revising the AWL section. The MCAI states: